



ERRATA

Aircraft System Identification: Theory and Practice

by Vladislav Klein and Eugene A. Morelli

Known errors are listed below. Please report any other errors to e.a.morelli@nasa.gov.

Page	Correction
x	Line 5, (GLS) should be deleted.
37	<p>The rotation matrices in Eq. (3.27) are incorrect. Eq. (3.27) should be:</p> $\begin{bmatrix} \dot{x}_E \\ \dot{y}_E \\ \dot{z}_E \end{bmatrix} = \begin{bmatrix} \cos\psi & -\sin\psi & 0 \\ \sin\psi & \cos\psi & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} \cos\theta & 0 & \sin\theta \\ 0 & 1 & 0 \\ -\sin\theta & 0 & \cos\theta \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos\phi & -\sin\phi \\ 0 & \sin\phi & \cos\phi \end{bmatrix} \begin{bmatrix} u \\ v \\ w \end{bmatrix}$
63	<p>At the far right of the first line in Eq. (3.110a), ΔC_D should be deleted. Eq. (3.110a) should be:</p> $\dot{V} = -\frac{\bar{q}_o S}{m} \left(C_{D_V} \frac{\Delta V}{V_o} + C_{D_\alpha} \Delta\alpha + C_{D_q} \frac{q\bar{c}}{2V_o} + C_{D_\delta} \Delta\delta \right) - g \cos\gamma_o (\Delta\theta - \Delta\alpha) - \frac{T_o \sin\alpha_o}{m} \Delta\alpha$
85	<p>There should be a transpose on the second Φ on the right side of Eq. (4.47). Eq. (4.47) should be:</p> $P(i i-1) = \Phi(i-1)P(i-1 i-1)\Phi^T(i-1) + \Gamma_w(i-1)Q(i-1)\Gamma_w^T(i-1)$
117	<p>In the first sentence of the last paragraph, the word “variables” should be inserted after “independent”. The first sentence of the last paragraph should be:</p> <p>For all flight test data sets and many wind tunnel data sets, the measured values of the independent variables are not uniformly spaced over an interval.</p>

Page	Correction																														
144	<p>The subscript for ν on the left side of Eq. (5.135) should be “z”, not “2”. Eq. (5.135) should be:</p> $\nu_z(i) = z(i) - \hat{\theta}_0 - \hat{\theta}_1 \xi_1(i) \quad i = 1, 2, \dots, N$																														
250	<p>Some of the numbers in Table 7.3 are incorrect. Table 7.3 should be:</p> <p>Table 7.3 Parameter estimation results for Schroeder sweep forced oscillation on an F-16XL 2.5 percent model</p> <table> <tr> <th></th> <th colspan="2">Frequency-domain equation-error</th> <th colspan="2">Frequency-domain output-error</th> </tr> <tr> <th>Parameter</th> <th>$\hat{\theta}$</th> <th>$s(\hat{\theta})$</th> <th>$\hat{\theta}$</th> <th>$s(\hat{\theta})$</th> </tr> <tr> <td>A</td> <td>8.58×10^{-1}</td> <td>3.25×10^{-2}</td> <td>8.32×10^{-1}</td> <td>3.44×10^{-2}</td> </tr> <tr> <td>B</td> <td>3.22×10^0</td> <td>3.84×10^{-2}</td> <td>3.21×10^0</td> <td>3.17×10^{-2}</td> </tr> <tr> <td>C</td> <td>1.82×10^{-1}</td> <td>7.05×10^{-2}</td> <td>1.95×10^{-1}</td> <td>2.80×10^{-2}</td> </tr> <tr> <td>b_1</td> <td>1.46×10^{-1}</td> <td>2.64×10^{-2}</td> <td>1.55×10^{-1}</td> <td>1.19×10^{-2}</td> </tr> </table>		Frequency-domain equation-error		Frequency-domain output-error		Parameter	$\hat{\theta}$	$s(\hat{\theta})$	$\hat{\theta}$	$s(\hat{\theta})$	A	8.58×10^{-1}	3.25×10^{-2}	8.32×10^{-1}	3.44×10^{-2}	B	3.22×10^0	3.84×10^{-2}	3.21×10^0	3.17×10^{-2}	C	1.82×10^{-1}	7.05×10^{-2}	1.95×10^{-1}	2.80×10^{-2}	b_1	1.46×10^{-1}	2.64×10^{-2}	1.55×10^{-1}	1.19×10^{-2}
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379	<p>There should not be a power of 2 applied to N in the denominator on the right side of Eq. (11.66). Eq. (11.66) should be:</p> $\frac{1}{N} \sum_{i=0}^{N-1} u^2(i) = \frac{1}{N \sum_{i=0}^{N-1} w^2(i)} \sum_{k=0}^{N-1} U_w(k) U_w^*(k)$																														